The Principle and Application of Maser Navigation

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Abstract

The traditional celestial navigation system (CNS) is used the moon, stars, and planets as celestial guides. Then the star tracker (i.e. track one star or planet or angle between it) and star sensor (i.e. sense many star simultaneous) be used to determine the attitude of the spacecraft. Pulsar navigation also be introduced to CNS. Maser is another interested celestial in radio astronomy which has strong flux density as spectral line. Now we analysis the principle of maser navigation which base on measuring Doppler shift frequency spectra and the feasibility that use the exist instrument. We give the navigation equations of maser-based navigation system and discuss the integrated navigation use maser, then give the perspective in the Milky Way and the intergalatic. Our analysis show that use one meter antenna can achieve tens of meters position accuracy which better than today's star sensor. After integrated with maser navigation, pulsar navigation and star sensor in CNS and inertial navigation system, is it not only increase the reliability and redundancy of navigation or guiding system but also can less or abolish the depend of Global Navigation Satellite System (GNSS) which include GPS, GRONSS, Galileo and BeiDou et al. Maser navigation can give the continuous position in deep space, that means we can freedom fly successfully in the Milky Way which use celestial navigation that include maser, pulsar and traditional star sensor. Maser as nature beacon in the universe will make human freely fly in the space of the Milky Way, even outer of it. That is extraordinary in the human evolution to type III of Kardashev civilizations

1 Introduction

Maser is a device that produces coherent electromagnetic waves through amplification due to stimulated emission. Historically the term came from the acronym "Microwave Amplification by Stimulated Emission of Radiation", although modern masers emit over a broad portion of the electromagnetic spectrum. Astrophysical maser is a naturally occurring source of stimulated spectral line emission, typically

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in the microwave portion of the electromagnetic spectrum. It was discovered by Weaver, H. et al. firstly (Weaver et al., 1965), after Charles Townes given prediction and Weinreb, S. et al. firstly detected the hydroxyl molecule (OH) (Weinreb, 1963), that was the first radio observation of an interstellar molecule. This emission may arise in molecular clouds, comets, planetary atmospheres, stellar atmospheres, or from various conditions in interstellar space.

The traditional CNS origin from nautical, developed to aeronautics by US (B52, B-1B, B-2A, C-141A, SR-71, F22 et al.) and Soviet (TU-16, TU-95, TU-160 et al.) (Pappalardi et al., 2001; AnGuo, 2007), success in determine the attitude of the spacecraft in help orient the Apollo spacecraft enroute to and from the Moon. Although the GNSS and Inertial Navigation System (INS) almost can finish any job in this planet now, someone still continued think it is important for it can be used independently of ground aids and has global coverage, it cannot be jammed (except by clouds) and does not give off any signals that could be detected by an the others. The traditional maritime state which include US, Russia, UK, French, all spend many money in CNS for its unique advantage.

Pulsar navigation is use pulsar as beacon give the continuous position in deep space. Dr Sheikh et al. construct the X-ray pulsar-based autonomous navigation theory which based modern spacecraft navigation technique that include Kalman filter et al. (Sheikh, 2005; Sheikh et al., 2007). Dong Jiang analysis the feasibility that use radio pulsar navigation and discuss the integrated navigation use pulsar, then give the different navigation mission analysis and design process basically which include

the space, the airborne, the ship and the land of the planet or the lunar in the solar system (Dong, 2008). With the distance increase, the radiometric tracking of deep space network (DSN) will decrease in accuracy, and it can not work when spacecraft in the other side of sun (Ray et al., 2008) and land rover in the back of the other planet or lunar. But pulsar can not be effected in that place.

A maser-based navigational system is considered by Shapiro et al. using the emissions from H₂O molecules which are the most intense in Very Long Baseline Interferometry (VLBI) navigation (Shapiro, Uliana and Yaplee, 1972). The VLBI technique, with a master station, can use either an artificial satellite or natural sources as position references, a highspeed data link is required. The characteristics of natural radio sources, their flux, distribution on the sky, and apparent size are shown to provide a limit on position measurement precision (Knowles and Johnston, 1973). Then Wallace, K. discuss that use radio sextant and radio star which include maser to navigation (Wallace, 1988) that just is the geometry method of traditional nautical celestial navigation. They thought the accuracy of a "radio sextant" is dependent amongst other things on the signal bandwidth, and the line emission which is typically in the region of 50 kHz is too narrow to attain reasonable fix accuracies (< 10 nm) without the use of long-baseline interferometric techniques.

Now I analysis the principle of maser autonomous navigation system which base measure Doppler shift frequency spectra in one receiver that is the similar process of measure Doppler effect in radio navigation, astrody-

namic and spacecraft navigation technique.

2 Principle of Maser Navigation

2.1 Doppler effect in Maser Observation

Maser emission from molecules such as water, hydroxyl (OH), and silicon monoxide (SiO) is strong spectral line that is an important tracer of the gas kinematics and magnetic field strength in astrophysical interesting regions. Figure. 1 show some examples of spectra from maser in Post-AGB stars (Deacon, Green and Chapman, 2004). The order of velocity is dozens of kilometers per second in this figure. Some of it have double peaks structure for the Doppler effect that come from the rotation of star.

The Doppler effect (or Doppler shift) is the change in frequency and wavelength of a wave for an observer moving relative to the source of the waves. It is commonly heard when a vehicle sounding a siren approaches, passes and recedes from an observer. The received frequency is increased (compared to the emitted frequency) during the approach, it is identical at the instant of passing by, and it is decreased during the recession. For waves which do not require a medium, such as light or gravity in special relativity, only the relative difference in velocity between the observer and the source needs to be considered. The Doppler effect for electromagnetic waves such as light is of great use in astronomy and results in either a so-called redshift or blueshift. It has been used to measure the

speed at which stars and galaxies are approaching or receding from us, that is, the radial velocity. This is used to detect if an apparently single star is, in reality, a close binary and even to measure the rotational speed of stars and galaxies.

According to the relativistic Doppler effect, we will have the relation between the frequency we will receive f' and the frequency the source emission f_0 :

$$\frac{f'}{f_0} = \frac{\sqrt{1 - \beta^2}}{1 - \beta * \cos\theta} \;,$$

here $\beta=\frac{\nu}{c}$, ν is the relative velocity between the source and the observer, θ is the angle between the line of the source to the observer and the direction of the source movement, c is the speed of light.

When $\theta = 90^{\circ}$, called transverse Doppler effect, we have the relation:

$$\frac{f'}{f_0} = \sqrt{1 - \beta^2} \ .$$

When $\theta=0^\circ$ and $\theta=180^\circ$, called longitudinal Doppler effect, we have the relation:

$$\frac{f'}{f_0} = \sqrt{\frac{1+\beta}{1-\beta}} \;,$$

and

$$\frac{f'}{f_0} = \sqrt{\frac{1-\beta}{1+\beta}} \ .$$

In usual, the transverse Doppler effect far less than longitudinal Doppler effect, so astronomer only calculate longitudinal Doppler effect. When $\nu \ll c$, we have:

$$\frac{f'}{f_0} = 1^+_- \beta$$
.

So the value of Doppler shift spectra

$$\Delta f = f' - f_0 = {}^{+}f_0\beta$$
 . (1)

In the formula, "+", "-" correspond to blueshift and redshift.

From the above formula, if we know the frequency f' and the frequency the source emission f_0 , we will have the relative velocity between the source and the observer. In astronomical observation, the velocity be normalized to the local standard of rest (LSR) for it benefit to study the celestial in a uniform frame. So I think we can use the Doppler effect to navigation. Figure. 2 show the principle of maser navigation in two dimension. The center is LSR, V is the velocity of spacecraft. If we can receive two signal which come from maser sources, we will have the relative velocity between the observer (i.e. the vehicle) and the LSR. Then using the velocity plus the time, we will have the relative position between the vehicle and the LSR. The similar principle of maser navigation in three dimension, we will have the information of the continuous position in the space, if we can receive three maser signal simultaneous.

2.2 Kalman filter for Maser Navigation

The kalman filter is an efficient recursive linear filter that estimates the state of a dynamic system from a series of noisy measurements (Kalman, 1960). It is mainly used to estimate system states that can only be observed indirectly or inaccurately by the system itself. It can predict the motion of anything for it is recursive, even the signal have noise for that use the dy-

namic state estimate the system. In maser navigation, that is significant like it in INS and the traditional CNS (i.e. star sensor). We can use navigation kalman filter measure spectral line range, spacecraft clock, then compare with the signal which come from maser, so we will have the velocity and position through plus time.

In order to use the kalman filter, one must model the process in accordance with the framework of the kalman filter. This means specifying the following matrices: A_k , the state-transition model; H_k , the observation model; Q_k , the covariance of the process noise; R_k , the covariance of the observation noise; and sometimes B_k , the control-input model, for each time-step, k, as described below. The kalman filter model assumes the true state at time k is evolved from the state at (k-1) according to

$$\mathbf{x}_k = \mathbf{A}_k \mathbf{x}_{k-1} + \mathbf{B}_k \mathbf{u}_k + \mathbf{w}_k , \qquad (2)$$

where A_k is applied to the previous state \mathbf{x}_{k1} ; \mathbf{B}_k is applied to the control vector \mathbf{u}_k ; \mathbf{w}_k is the process noise which is assumed to be drawn from a zero mean multivariate normal distribution with covariance \mathbf{Q}_k , $\mathbf{w}_k \sim \mathbf{N}(0, \mathbf{Q}_k)$. At time k an observation (or measurement) \mathbf{z}_k of the true state \mathbf{x}_k is made according to

$$\mathbf{z}_k = \mathbf{H}_k \mathbf{x}_k + \mathbf{v}_k \,, \tag{3}$$

where \mathbf{H}_k maps the true state space into the observed space and \mathbf{v}_k is the observation noise which is assumed to be zero mean Gaussian white noise with covariance \mathbf{R}_k , $\mathbf{v}_k \sim \mathbf{N}(0,\mathbf{R}_k)$. The initial state, and the noise vectors at each step $x_0,w_1,...,w_k,v_1...v_k$ are all assumed to be mutually independent.

Figure 3 show the system model of the (linear) kalman filter. At each time step the state

vector \mathbf{x}_k is propagated to the new state esti- Here the weight of the measurement x_i is p_i , mation \mathbf{x}_{k+1} by multiplication with the constant state transition matrix A. The state vector \mathbf{x}_{k+1} is additionally influenced by the control input vector \mathbf{u}_{k+1} multiplied by the input matrix \mathbf{B} , and the system noise vector \mathbf{w}_{k+1} . The system state cannot be measured directly. The measurement vector \mathbf{z}_k consists of the information contained within the state vector \mathbf{x}_k multiplied by the measurement matrix H, and the additional measurement noise \mathbf{v}_k .

2.3 Navigation Equations of Maserbased Navigation System

The receiver uses messages received from three masers to determine the telescope positions. The x, y, and z components of velocity sent are designated as $[V_x, V_y, V_z]$, where the subscript i denotes the masers and has the value 1, 2 or 3. So we have navigation equations:

$$\vec{V} = V_x^2 + V_y^2 + V_z^2 \,. \tag{4}$$

Here, the direct measurements is velocity which relative to masers.

We will have the instantaneous change of the position that use the instantaneous velocity multiplied with time, the position equation is:

$$\delta S = \delta \vec{V} \times \delta t \ . \tag{5}$$

If we only can observe two or one maser, we still can perform maser navigation in combination with the orbit of the mobile station that use kalman filter et al. If it is two masers, we can give the weighted mean of position and time from the equation:

$$x^* = \frac{\sum p_i x_i}{\sum p_i} \,. \tag{6}$$

 $p_i = \sigma_1^2/\sigma_i^2$ $(i = 1 \sim N), \sigma_i$ the standard deviation of unequal precision measurement sequence, we set $\sigma_1 = max\sigma_i$ $(i = 1 \sim N)$ in usual.

Maser signal process in astronomy Vs The requires of engineer project Vs The reliable of technique

Maser have been found in transitions of OH, SiO, water, methanol, ammonia, and other molecules, and also in recombination lines of hydrogen. The maser observation system sensitivity (i.e. the raw limiting flux density) is given by the radiometer equation:

$$S_{\text{lim}} = \frac{\sigma \beta}{(BN_{\text{p}}\tau_{obs})^{1/2}} \frac{T_{\text{sys}}}{G} , \qquad (7)$$

where σ is a loss factor, taken to be 1.5 (One-bit sampling at the Nyquist rate introduces a loss of $\sqrt{2/\pi}$ relative to a fully sampled signal. The principal remaining loss results from the nonrectangular bandpass of the channel filters). β is the detection signal-to-noise ratio threshold, taken to be 5.0, G is the telescope gain, B is the receiver bandwidth in Hz, $N_{\rm p}$ is the number of polarizations and $\tau_{\rm obs}$ is the time per observation in seconds. $T_{\rm sys}$ is the system temperature, G is the telescope gain, $G = A_e/(2k_B)$, here A_e is the effective area of a telescope, $k_{\rm B}$ is Boltzmann's constant.

Using the above formula, we use 4 M antenna (If the telescope efficiency is 0.8, $A_{\rm e} = 0.8 \times$ $\pi (4/2)^2 \simeq 10 \text{ m}^2, G \simeq 3.62 \times 10^{-3} \text{ KJy}^{-1}$, worth in dBm ? $P_{\rm mW} = 10^{-26} \text{ W/m}^2/\text{Hz} \times 10^{-2} \text{ M/m}^2$ we set $T_{\rm sys}$ is 20 K, B is 1 MHz, $N_{\rm p}$ is 2, 1,000,000 Hz \times 12.6 m² \times 1000 mW/W = $\tau_{\rm obs}$ is 4 min, λ_0 is 22.2 GHz (H₂O Interstellar Maser), so we have $S_{\rm lim} \simeq 1.89$ Jy. If we set B is 10 kHz, $\tau_{\rm obs}$ is 1 sec, the other is not change, we have $S_{\rm lim} \simeq 293$ Jy. The flux density of maser is very biggest, The table 1 is the type list of the strong radio maser source, we can observed it even if use 4 meter antenna in microwave in one second. If the diameter of dish is 1 M (If the telescope efficiency is $0.8, A_e = 0.8 \times \pi (1/2)^2 \simeq 0.68 \times 10^{-1} \text{ m}^2,$ $G \simeq 2.27 \times 10^{-4} \text{ KJy}^{-1}$), we set T_{sys} is 20 K, B is 10 kHz , $N_{\rm p}$ is 2, $\tau_{\rm obs}$ is 4 sec, λ_0 is 22.2 GHz (H₂O Interstellar Maser), so we have $S_{\rm lim} \simeq 2.34 \times 10^3 \, \rm Jy.$ From the table 1, we Still can use H₂O maser to navigation.

The above formula use Jy as unit. The Jansky (Jy) is a measure of spectral power flux density - the amount of RF energy per unit time per unit area per unit bandwidth, 1 Jy \equiv $10^{-26} \text{ W/m}^2/\text{Hz}$. The jansky is not used outside of radio astronomy. It is not a practical unit for measuring communications signals, the magnitude is much too small, and is a linear unit, very few RF engineers outside of radio astronomy will know what a Jy is. Because of wide dynamic range encountered the most radio systems, the power is usually expressed in logarithmic units of watts (dBW) or milliwatts (dBm): $dBW \equiv 10 log_{10} Power_{watts}$, $dBm \equiv 10log_{10}Power_{milliwatts}$. While not comprised of the same units, we can make some reasonable assumptions to compare a Jy to dBm. Assumptions bandwidth is 1 MHz, 22.2 GHz frequency ($\lambda_0 = 0.01 \text{ m}$), parabolic receive antenna, antenna collecting area = $\pi \times r^2$ =

 $1.26 \times 10^{-16} \text{ mW}, P_{\text{dBm}} = 10 log(1.26 \times 10^{-16})$ 10^{-16} mW) = -158.9963 dBm. Considering the parabolic antenna as a circular aperture gives the following approximation for the maximum gain: $G_{\rm dBi} \simeq 10 log((9.87 \times D^2)/\lambda_0^2)$. in this form, G is power gain over isotropic D is reflector diameter in same units as wavelength, λ_0 is center of wavelength. For 4 M diameter and $\lambda_0 = 0.013 \text{ m}, G_{\text{dBi}} = 59.3777. \text{ So } 1 \text{ Jy in 4 M}$ antenna is -99.6186 dBm. If we set 10^3 Jy , we have $P_{\rm dBm} = 128.9963 \; \rm dBm$, after 4 M antenna amplify it, we have the signal is 69.6186 dBm.

Masers take place in several places in the universe: in the vicinity of newly forming stars and regions of ionized hydrogen (OH, water, SiO, and methanol masers); in the circumstellar shells of stars at the end of its life that is, red giants and supergiants (OH, water, and SiO masers); in the shocked regions where supernova remnants are expanding into an adjacent molecular cloud (OH masers); and in the nuclei and jets of active galaxies (OH and water masers). The emission from OH masers can vary on timescales of hundreds of seconds and be detected as long-duration radio bursts (Cohen and Brebner, 1985; Yudaeva, 1986). In the above, maser from circumstellar matter of red giants and supergiants (i.e. AGB and post-AGB) is well in navigation for some of it have double peaks structure that easily identified. The emission from maser of circumstellar matter have vary on timescales of orders of three months to years (Etoka et al., 2001; Lekht et al., 2001).

Navigation of use maser just for a continu- $3.14 \times (4/2)^2 = 12.6 \text{ m}^2$. How much is one Jy ous spectral line signal during the different mission time which during tens of minutes to several years. When we penetrate the system of maser navigation as one systems engineering, I think navigation system use maser is feasible absolutely. In maser navigation, radial velocity measurements and the time measurements is important to have the position. The accuracy of this navigation system only depends on the accuracy of the spectrum we have obtained. From equation 7, we have $V = \Delta f \times c/f_0$, so 1 kHz shift in 22.2 GHz corresponds to about 13.5 ms^{-1} . Today these spectrometers (autocorrelators, acusto-optical spectrometers and filterbank) offer a useable bandwidth from a few kHz up to 2 GHz with a few thousand spectral channels, which are capable of resolving narrow spectral lines of masers. If we can measure 1 Hz shift of H₂O maser, we will have the velocity accuracy is $1.35 \, \mathrm{cm s^{-1}}$. The wide bandwidth of receiver will give the big measuring range of velocity. It also is important to maser navigation.

The light frequency comb have developed in recently (Hall, 2006; Hänsch, 2006) that will play key role in maser navigation. A laser frequency comb that enables radial velocity measurements with a precision of 1 cms^{-1} (Li et al., 2008). If we can achieve the similar instrument in microwave, we can have easily finish maser navigation. The atomic clock has the advantage that keep time in short timescale. Pulsar especially millisecond pulsars (MSP) be thought the natures most stable clocks (Taylor, 1991). The data show some pulsar stability than atomic clock in timescale than one year (Matsakis, Taylor and Eubanks, 1997). When integrated it, even plus light frequency comb clock in the future, that will satisfied with maser navigation. Some modern digital signal processing (DSP) technique can apply to maser signal navigation which include weak signal detection, signal enhancement, signal reconstruct et al. Maser spectral line is Gaussian for the thermal motion of molecule, but the complex surrounding for example turbulence make profile become complex. In navigation, we just need the information from phase, so we can magnify the weak maser profile signal through normalizing it to a Gaussian signal or plus a Gaussian signal. The navigation system must leave a copy of raw data to astronomer for the best filter is construct a good noise model by it.

Dong Jiang analysis the special parabolic dish use in spacecraft to achieve pulsar tracker, the phased array antenna to achieve pulsar sensor, and the phased array feed can apply in pulsar sensor when use one dish (Dong, 2008). The similar technique can use to maser sensor in navigation. The phased array antenna or feed can receive several radio celestial which include different maser or pulsar simultaneous. With electronic technique development, high speed analog-to-digital converter (ADC) obtain order of Gigabyte^{-s} easily, field-programmable gate array (FPGA), multi-core multi-PC cluster and graphics processing unit (GPU) all apply to scientific computing. If we can fuse Multi-core CPU, GPU and FPGA, construct one computing server and use different advantage of it, That will easily finish many scientific computation which include reduce different radio sources.

Integrated navigation with maser between pulsar navigation, CNS and INS, even GNSS, is realistic path in the future mission. It will increase the reliability and redundancy of navigation or guiding system (Zhang and Zeng, 2008). The multi-waveband maser navigation also is in-

terested. In integrated navigation, system analysis and modeling, system state estimation, filter design, information synchronization and system fault tolerance filter design all is important. Dong Jiang give the different navigation mission analysis and design process basically which include the space, the airborne, the ship and the land of the planet or the lunar in the solar system (Dong, 2008). The similar analysis also fit for maser navigation.

4 Maser Navigation In the Milky Way and Intergalaxy

The virtue is obvious, when the rover in the back of the others planet or lunar, DSN can not work and human can not built GNSS for the other planet in long term. So the maser navigation and radio pulsar navigation is one and only method at any place of the other planet surface day and night in the future explore. The advantage of maser navigation is some of beacon that maser emission come from the nuclei and jets of active galaxies (OH and water masers) (Lo, 2005). So human will have chance use it freely fly in the space of the Milky Way and Intergalaxy.

Soviet astronomer Kardashev N. S. proposed a scheme for classifying advanced technological civilizations. He identified three possible types and distinguished between them in terms of the power they could muster for the purposes of interstellar communications. The Type III civilization would have evolved far enough to tap the energy resources of an entire galaxy. This would give a further increase by at least a factor of 10 billion to about 10^{36} watts (Kardashev,

1964). If we want to use the resources of someplace, we must freely voyage in the that space firstly. Maser navigation must be extraordinary in the human evolution to type III of Kardashev civilizations.

Now the star sensor in optic can give the position accuracy is $100 \text{ m} (3\sigma)$. Digital spectrum analyzer can provide the resolution is 100 Hz in frequency domain in today, so it can provide the accuracy about 1.35 ms^{-1} in velocity and the position accuracy is several meters. This means that maser navigation is better in the position accuracy than star sensor which does not require any technological breakthroughs. In order to get better result from maser navigation system, we still need the biggest telescope find the more masers, and the special telescope study maser's noise and astrometric model, and the corresponding software and hardware to improve the spectral lines shift estimater of maser.

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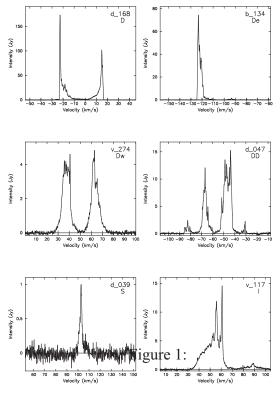
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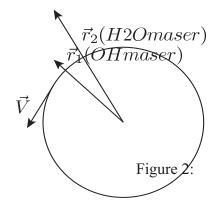


Fig. 1. Some examples of spectra from maser in Post-AGB Stars(Deacon, Green and Chapman, 2004).

Fig. 2. The principle of maser navigation in two dimension.

Fig. 3. This is the system model of the (linear) kalman filter. The picture download from net, author is Burkart Lingner.

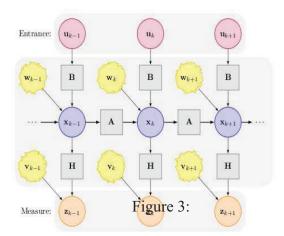


Table 1: The classical parameter of the strong Masers.

| Name | F | T_{B} | $\Delta \nu / \delta \nu$ |
|--|-------------------|------------------|---|
| Maser class | (Jy) | (K) | $(\text{km} * \text{s}^{-1}/\text{km} * \text{s}^{-1})$ |
| OH-Stellar Maser (1612 MHz) | 2×10^{2} | 10^{8} | 1/30 |
| H ₂ O-Stellar Maser (22.2 GHz) | 4×10^{3} | 10^{11} | 1/20 |
| SiO-Stellar Maser (43.1 GHz) | 2×10^{2} | 10^{10} | 1/10 |
| OH-Interstellar Maser (1665 MHz) | 2×10^{2} | 10^{12} | 0.1/50 |
| H ₂ O-Interstellar Maser (22.2 GHz) | 4×10^{3} | 10^{14} | 1/50 |
| CH ₃ OH-Interstellar Maser (12.1 GHz) | 5×10^{2} | $> 10^{12}$ | 1/5 |

F is flux density, $T_{\rm B}$ is the brightness temperature, $\Delta \nu/\delta \nu$ is the ratio of the velocity width of maser and the velocity range of the whole maser.